XtraExport

Queen Bee Air Specialties inc.

136 N Yellowstone Hwy.

Rigby Id 83442

United States

FAA Airworthiness Directives Compliance Record

Company:

Teton West

Aircraft Registration No: N2507S

Category: Airframe

Position:

ATP Revision: 4/7/2023

Manufacturer: Textron Aviation Inc.

P/N: LC41-550FG

Model: Columbia 400

S/N: 41567

| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
|---|--|----------------|---|-------|------------------|--|
| 2022-03-15 3/21/2022 | To Ensure That the Amount of Fuel Indicated is the Amount of Fuel Available. The Unsafe Condition, if Not Addressed, Could Result in Fuel Starvation and Engine Shutdown Which Could Result in the Inability to Arrive at the Destination Airport or, contd. | Hrs: C: | | No | D: Hrs: C: | |
| 2021-23-12 12/9/2021 | To Address the Radio Altimeter Anomalies That Are Undetected by the Automation or Pilot, Particularly Close to the Ground (e.g., Landing Flare), Could Lead to Loss of Continued Safe Flight and Landing | Hrs: C: | N/A - Affected item is not installed in the aircraft as of March 2022 | No | D: Hrs: C: | |
| 2013-11-10 7/26/2013 | To ensure the proper use of main landing gear brakes | Hrs: C: | PCW August 6, 2013 by inspection. See POH for an additional documentation noted in the log | No | D: Hrs: C: | |
| 2011-03-04 3/14/2011 | [Recurring] To detect and correct damage, i.e., cracking, deformation, and discoloration, in the rudder hinges and the,contd. | Hrs: C: | PCW Kit 09-27-01 was installed June 28, 2010. | No | D: Hrs: C: | |

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Phone: 2087457654

| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
|---|---|------------------------|--|-------|----------------------------|--|
| 2010-26-54 1/10/2011 | To prevent catastrophic failure of the wing due to disbonding of the wing skin from the wing spar | Hrs: C: | N/A S/N | No | D: Hrs: C: | |
| 2010-26-53 E 12/10/2010 | Superseded by 2010-26-54 | Hrs: C: | | No | D: Hrs: C: | / |
| 2009-09-09 5/11/2009 | [Recurring] Superseded by 2011-03-04 | Hrs: C: | | No | D: Hrs: C: | / |
| 2007-07-06 R1 12/5/2008 | [Recurring] To prevent jamming in the nileron and elevator control systems, which could result in failure | 4/7/2023 Hrs: C: | Complied with by Inspection per Cessna SB07-002D e(2) and e(4). Inserted Appendix A of the SB into Limitations section of the POH. | | D: 4/30/2024 Hrs: C: | QBAS 3470895/IA Charles Kemper |
| 2006-25-08 12/21/2006 | To prevent a short circuit condition at the deice heater connector, which could result in damage to the wings, contd. | Hrs: C: | N/A S/N | No | D: Hrs: C: | / |
| 2005-12-20 C2 6/21/2005 | [Recurring] To detect and correct cracks in the elevator torque tube assembly, which could result in failure of the, contd. | Hrs: C: | N/A S/N | No | D: Hrs: C: | / |
| | | Danisia | | | | |

Category: Engine Position: ATP Revision: 4/7/2023

P/N: TSIO-550-C20B Manufacturer: Teledyne Continental

S/N: 1008794 Model: TSIO-550-C

| Issue Number Effective Date Amendment # Descr | ription Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
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| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
|---|---|-------------------------------|---|-------|------------------|---|
| 2023-05-16 3/15/2023 | To Prevent Departure of Counterweight and Retaining Hardware from the Crankshaft Assembly. The Unsafe Condition, if Not Addressed, Could Result in Loss of Engine Oil Pressure, Catastrophic Engine Damage, Engine Seizure, and Consequent Loss of the Aircraft | 4/7/2023 Hrs: 2200.5 C: | Removed the #1 & # cyilinders and inspected counterweight attachment pins and snap-rings IAW Paragraph G and Continental Mandatory Service Bulletin MSB23-01, Revision A. Proper engagement of each snap-ring was inspected then measured with the provided tool/gauge. No defects noted. | 3No | D: Hrs: C: | QBAS 3470895/A&P/IA Charles Kemper |
| 2023-04-08 2/23/2023 | Superseded by 2023-05-16 | Hrs: C: | | No | D: Hrs: C: | |
| 2023-02-12 2/17/2023 | To Prevent Failure of the Engine Intake Valve. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Engine, In-Flight Shutdown, and Loss of the Airplane | Hrs: C: | N/A - The affected items are not installed. | No | D: Hrs: C: | |
| 2020-16-11 9/21/2020 | To Prevent Failure of the Engine. If Not Addressed, Could Result in Failure of the Engine, In-Flight Shutdown, and Forced Landing | Hrs: C: | N/A Overhaul Date and lack of cylinder changes | No | D: Hrs: C: | |
| 2016-21-04 11/18/2016 | | Hrs: C: | PCW June 6, 2017. Fitting replaced | No | D: Hrs: C: | / |
| 2016-16-12 9/15/2016 | To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight, contd. | Hrs: C: | N/A Effected Part not installed | No | D: Hrs: C: | / |

| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
|---|--|--|--|-------|------------------|---|
| 2012-03-06 C 2/24/2012 | To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane | The same of the sa | N/A Not Installed | No | D: Hrs: C: | |
| 2011-13-03 7/13/2011 | To prevent seizure of the turbocharger turbine, which could result in damage to the engine, & smoke in the, contd. | Hrs: C: | N/A Turbo's overhauled since the date of AD issue. | No | D: Hrs: C: | |
| 2010-11-04 6/16/2010 | To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane | Hrs: C: | N/A Engine Overhaul Date | No | D: Hrs: C: | / |
| 2009-24-52 E 11/18/2009 | Superseded by 2010-11-04 | Hrs: C: | | No | D: Hrs: C: | / |
| 2007-16-10 8/23/2007 | To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. | Hrs: C: | N/A Turbo's overhauled since the date of AD issue. | No | D: Hrs: C: | |
| 2004-08-10 5/5/2004 | To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. | Hrs: C: | N/A Effected Cylinders not installed | No | D: Hrs: C: | |
| 00-00-01 1/22/2001 | Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 | Hrs: C: | N/A Adapter not installed | No | D: Hrs: C: | / |
| 2000-23-21 12/12/2000 | To prevent crankshaft connecting rod journal fracture, which could result in total engine power,contd. | Hrs: C: | N/A Date of Engine Overhaul | No | D: Hrs: C: | / |
| 2000-08-51 E 4/28/2000 | Superseded by 2000-23-21 | Hrs: – C: | Ang. | No | D: Hrs: C: | |

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| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
|---|--|----------------|---|-------|------------------|---|
| 99-19-01 9/30/1999 | To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. | Hrs: C: | N/A Date of Engine Overhaul | No | D: Hrs: C: | |
| 99-09-17 L 4/22/1999 | Superseded by 99-19-01 | Hrs: C: | | No | D: Hrs: C: | |
| 96-12-22 7/31/1996 | [Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD. | Hrs: C: | N/A Oil Filter Adapter Not Installed | No | D: Hrs: C: | • |
| 93-10-02 8/12/1993 | TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY | Hrs: C: | N/A Date of Engine Overhaul | No | D: Hrs: C: | |

Category: Propeller Position: ATP Revision: 4/7/2023

Manufacturer: Hartzell Propeller

Model: HC-H3YF-1

S/N: LX149B

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|---|--|----------------|---|-------|------------------|--|
| 2007-26-09 1/30/2008 | To prevent failure of the propeller blade from fatigue cracks in the aluminum blade shank radius, which can,contd. | Hrs: C: | N/A Affected Propeller Blades are not Installed | No | D: Hrs: C: | |
| 2005-14-11 8/17/2005 | To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane | Hrs: C: | N/A SoCal was not utilized | No | D: Hrs: C: | / |
| 2002-09-08 5/13/2002 | Superseded by 2007-26-09 | Hrs: C: | | No | D: Hrs: C: | |

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|---|--|----------------|-------------------------|-------|------------------|---|
| 2001-07-03 C 6/4/2001 | To prevent propeller failure of the propellers returned to service by BASCO, & possible loss of airplane control | Hrs: C: | N/A BASCO not utilized | No | D: Hrs: C: | |
| 77-12-06 R(2) 12/21/1977 | [Recurring] Superseded by 2002-09-08 | Hrs: C: | | No | D: Hrs: C: | |
| 75-07-05 5/1/1977 | Superseded by 77-12-06 | Hrs: C: | | No | D: Hrs: C: | / |
| 74-15-02 1/1/1974 | Superseded by 77-12-06 | Hrs: C: | | No | D: Hrs: C: | |
| 73-10-03 1/1/1973 | Superseded by 77-12-06 | Hrs: C: | | No | D: Hrs: C: | / |
| 70-02-01 1/1/1970 | Superseded by 73-10-03 | Hrs: C: | | No | D: Hrs: C: | |
| 70-16-03 R 1/1/1970 | Superseded by 77-12-06 | Hrs: C: | | No | D: Hrs: C: | / |

Category: Magnetos

Position: Left

ATP Revision: 4/7/2023

Manufacturer: Continental Motors

P/N: BL-500556-101 S6RSC-25P

Model: S6-20 SERIES

S/N:

| Issue Number | A STATE OF THE STA | | 20.2 | | | Facility Cert No. / Type |
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| • | | Hrs: C: | | | D: Hrs: C: | |
| ategory: Magneto | os | Position | n: Right | | ATP Re- | vision: 4/7/2023 |

Manufacturer: Continental Motors

P/N: BL-500556-101 S6RSC-25P

Model: S6-20 SERIES

S/N:

| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
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| - | | Hrs: C: | | - | D: Hrs: C: | |

Category: Alternators Position: Belt ATP Revision: 4/7/2023

Manufacturer: Hartzell Engine Tech

Model: ALU-6526 Alternator

P/N:

S/N: H-0030640

Facility Issue Number Cert No. / Type Effective Date Method of Authorized By Amendment # Description Compliance Signed Complied Recur Next Due D: --Hrs: --Hrs: --

Category: Alternators Position: Direct Drive ATP Revision: 4/7/2023

Manufacturer: Hartzell Engine Tech

Model: 646843 Alternator

P/N:
S/N:

| Issue Number Effective Date | | 7 | Method of | | | Facility Cert No. / Type Authorized By |
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| Amendment # | Description | Complied | Compliance | Recur | Next Due | Signed |

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Category: Starter Position: ATP Revision: 4/7/2023

Manufacturer: Sky-Tec P/N:
Model: PL13484 S/N:

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| | | Hrs: C: | | - | D: Hrs: C: | |

Category: ELT Systems Position: ELT-200 ATP Revision: 4/7/2023

Manufacturer: Artex Aircraft Supplies

P/N: 453-0190

Model: Any Model

S/N: E08180

| Issue Number Effective Date Amendment # | Description | Complied | Method of Compliance | Recur | Next Due | Facility Cert No. / Type Authorized By Signed |
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| | | Hrs: C: | | - | D: Hrs: C: | |