

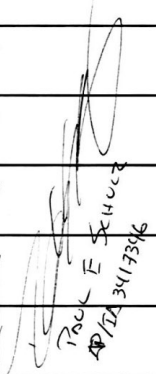
FAA Airworthiness Directives Compliance Record

Company: Buddyjack IIc
Category: Airframe
Manufacturer: Piper Aircraft, Inc.
Model: PA-32R-301T

Position:
P/N:
S/N: 3257454

Aircraft Registration No: N1717R
ATP Revision: 11/27/2019

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2017-14-04 08/15/2017	[Recurring] To prevent rupture or failure of the oil cooler hose assemblies, which could result in engine stoppage with,contd.	-- Hrs: -- C: --	N/A TSO-C53a, Type D hose	Yes	D: -- Hrs: -- C: --	
2013-02-13 03/11/2013	[Recurring] To prevent failures that may lead to failure of the horizontal stabilator control system and cold result,contd.	04/13/2007 Hrs: -- C: --	Not due at this time	Yes	D: 04/09/2022 Hrs: -- C: --	
2010-15-10 08/31/2010	To detect and correct any incorrectly assembled control wheel shafts	06/18/2014 Hrs: 516.4 C: --	PCW by inspection IAW SB 1197B	No	D: -- Hrs: -- C: --	
2010-13-07 C 07/28/2010	To prevent failure of the V-band exhaust coupling, which could cause the exhaust pipe to detach from the,contd.	-- Hrs: -- C: --		No	D: -- Hrs: -- C: --	
2004-14-12 C 08/10/2004	To detect & correct inadequate control wheel attachment design, which could result in loss of control of the,contd.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
96-10-03 06/14/1996	TO PREVENT FAILURE OF THE FLAP HANDLE ATTACH BOLT & SUDDEN RETRACTION OF THE FLAPS, WHICH, IF NOT DETECTED, CONTD.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	


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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
95-26-13 02/05/1996	[Recurring] Superseded by 2017-14-04	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
93-05-10 04/30/1993	[Recurring] TO PREVENT THE INABILITY TO RETRACT OR EXTEND THE NOSE LANDING GEAR, WHICH COULD RESULT IN SUBSTANTIAL, CONTD.	-- Hrs: -- C: --	N/A S/N	Yes	D: -- Hrs: -- C: --	
86-17-01 08/21/1986	TO PREVENT SMOKE IN THE COCKPIT AND POSSIBLY COMPLETE ELECTRICAL FAILURE RESULTING FROM SHORTING OF, CONTD.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
81-24-07 11/20/1981	TO PREVENT A POTENTIALLY HAZARDOUS CONDITION DURING LANDING AND GROUND OPERATION DUE TO THE FAILURE, CONTD.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
80-24-03 11/28/1980	Superseded by 86-17-01	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
80-14-03 07/01/1980	DISRUPTION OF RADIO COMMUNICATION	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
76-25-06 12/23/1976	Superseded by 95-26-13	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
72-04-03 01/01/1972	Superseded by 76-25-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
67-26-02 05/22/1968	VARIOUS MODIFICATONS	-- Hrs: -- C: --	N/A model	No	D: -- Hrs: -- C: --	

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
68-01-03 05/21/1968	FORWARD AND REAR WING-TO-FUSELAGE ATTACHMENT FITTINGS	-- Hrs: -- C: --	N/A application	No	D: -- Hrs: -- C: --	
67-20-04 09/27/1967	TO PRECLUDE THE POSSIBILITY OF FAILURE OF THE MAIN LANDING GEAR TORQUE LINK	-- Hrs: -- C: --	N/A part installed	No	D: -- Hrs: -- C: --	

Category: Engine

Position:

ATP Revision: 11/27/2019

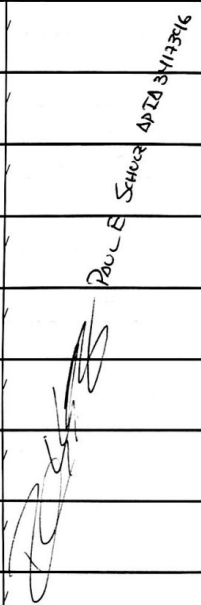
Manufacturer: Textron Lycoming

P/N:

Model: TIO-540-AH1A

S/N: L-12687-61

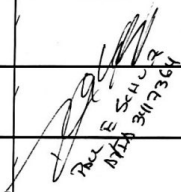
Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2017-16-11 08/15/2017	To prevent connecting rod failure which could result in uncontained engine failure, total engine power loss,contd.	-- Hrs: -- C: --	N/A S/N & affected work not completed during affected time	No	D: -- Hrs: -- C: --	
2015-19-07 11/03/2015	[Recurring] To prevent failure of the fuel injector fuel lines, which could lead to uncontrolled engine fire, engine,contd.	-- Hrs: -- C: --	Not due at this time. Next due at moment or adjustment of any injector line	Yes	D: -- Hrs: -- C: --	
2015-02-07 03/11/2015	To prevent the propeller governor shaft set screw from coming loose, causing damage to the engine and,contd.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
2012-19-01 10/24/2012	To prevent failure of the crankshaft, which will result in total engine power loss, in-flight engine,contd.	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
71-13-01 R1 03/08/2012	TO PREVENT POSSIBLE FAILURES OF THE FUEL INJECTOR MANIFOLD TO NOZZLE TUBE ASSEMBLIES	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2012-03-06 C 02/24/2012	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane	-- Hrs: -- C: --	N/A affected part not installed	No	D: -- Hrs: -- C: --	 Paul E Satchel AP ID 34113316
2011-26-04 01/25/2012	[Recurring] Superseded by 2015-19-07	-- Hrs: -- C: --	superseded	Yes	D: -- Hrs: -- C: --	
2011-15-10 08/16/2011	Superseded by 2012-03-06	-- Hrs: -- C: --	superseded	No	D: -- Hrs: -- C: --	
2011-13-03 07/13/2011	To prevent seizure of the turbocharger turbine, which could result in damage to the engine, & smoke in the,contd.	-- Hrs: -- C: --	N/A over 50 hrs TIS	No	D: -- Hrs: -- C: --	
2009-02-03 02/09/2009	[Recurring] To prevent a lean running engine, which could result in a substantial loss of engine power and subsequent,contd.	01/15/2019 Hrs: -- C: --	C/W at component O/H. NFA required	Yes	D: -- Hrs: -- C: --	
2008-14-07 08/14/2008	[Recurring] Superseded by 2011-26-04	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
2008-08-14 04/29/2008	[Recurring] Superseded by 2009-02-03	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
2008-06-51 E 03/12/2008	[Recurring] Superseded by 2008-08-14	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
2006-20-09 11/03/2006	Superseded by 2012-19-01	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-19-11 10/21/2005	To prevent failure of the crankshaft, which could result in total engine power loss, in-flight failure, and,contd.	-- Hrs: -- C: --	N/A part S/N	No	D: -- Hrs: -- C: --	
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	-- Hrs: -- C: --	N/A Slick mags installed	Yes	D: -- Hrs: -- C: --	
2004-10-14 C 06/25/2004	[Recurring] To prevent loosening or failure of the crankshaft gear retaining bolt, which may cause sudden engine failure	-- Hrs: -- C: --	not due until sudden stop of propeller	Yes	D: -- Hrs: -- C: --	
2004-05-24 C 03/30/2004	To prevent the loss of all engine power and possible forced landing	-- Hrs: -- C: --	N/A affected bolt p/n not installed at manufacture	No	D: -- Hrs: -- C: --	
2003-14-03 08/14/2003	[Recurring] To prevent rotary fuel pump leaks, which could result in an engine failure, engine fire, and damage to or,contd.	-- Hrs: -- C: --	N/A m suffix pump installed	Yes	D: -- Hrs: -- C: --	
2002-26-01 01/31/2003	[Recurring] Superseded by 2008-14-07	-- Hrs: -- C: --	Supperseded	Yes	D: -- Hrs: -- C: --	
2002-23-06 11/19/2002	Superseded by 2004-05-24	-- Hrs: -- C: --	Supperseded	No	D: -- Hrs: -- C: --	
2002-20-51 E 10/01/2002	Superseded by 2002-23-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
2002-19-03 09/20/2002	To prevent crankshaft failure, which could result in total engine power loss, in-flight engine failure and,contd.	-- Hrs: -- C: --	N/A STC not installed	No	D: -- Hrs: -- C: --	

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2002-17-53 E 08/16/2002	Superseded by 2002-19-03	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
2002-04-51 E 02/11/2002	Superseded by 2002-17-53	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
98-18-12 09/28/1998	[Recurring] Superseded by 2003-14-03	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
97-15-11 08/12/1997	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	
97-01-03 01/21/1997	Superseded by 97-15-11	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	
93-02-05 06/14/1993	[Recurring] Superseded by 2002-26-01	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
92-12-05 07/10/1992	TO PREVENT PISTON PIN FAILURE, OR PISTON RELEASE, AND ENGINE FAILURE	-- Hrs: -- C: --	C/W at engine manufacture	No	D: -- Hrs: -- C: --	
91-14-22 08/19/1991	[Recurring] Superseded by 2004-10-14	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
91-08-07 C 05/06/1991	TO PREVENT ENGINE POWER LOSS AND POSSIBLE LOSS OF THE AIRCRAFT	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
79-04-05 09/26/1979	TO PREVENT AN IN-FLIGHT POWER LOSS DUE TO THE SEPARATION OF THE P/N 2529192 REGULATOR DIAPHRAGM STEM ASSEMBLY	-- Hrs: -- C: --	N/A cw at manufacture	No	D: -- Hrs: -- C: --	
78-23-10 11/07/1978	TO PREVENT AN IN-FLIGHT POWER LOSS DUE TO AN OVER RICH CONDITION, CONTD.	-- Hrs: -- C: --	N/A cw at manufacture	No	D: -- Hrs: -- C: --	
75-09-15 04/30/1975	TO PREVENT POSSIBLE FUEL STARVATION TO THE ENGINE	-- Hrs: -- C: --	N/A affected part not installed	No	D: -- Hrs: -- C: --	

Category: Propeller

Position:

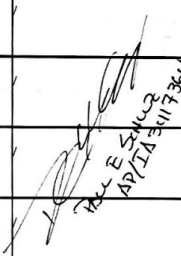
ATP Revision: 11/27/2019

Manufacturer: Hartzell Propeller

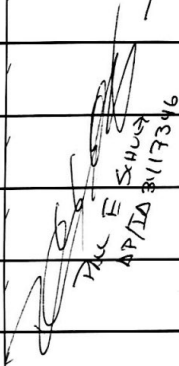
P/N:

Model: HC-13YR-1

S/N: HK1290B

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2007-26-09 01/30/2008	To prevent failure of the propeller blade from fatigue cracks in the aluminum blade shank radius, which can,contd.	-- Hrs: -- C: --	PCW at time of manufacture	No	D: -- Hrs: -- C: --	
2006-24-07 01/03/2007	To detect potentially unsafe conditions that could result in a propeller blade separating from the hub,contd.	-- Hrs: -- C: --	N/A RTS not by affected facility	No	D: -- Hrs: -- C: --	
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	-- Hrs: -- C: --	N/A RTS not by affected facility	No	D: -- Hrs: -- C: --	
2002-09-08 06/13/2002	Superseded by 2007-26-09	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2001-07-03 C 06/04/2001	To prevent propeller failure of the propellers returned to service by BASCO, & possible loss of airplane control	-- Hrs: -- C: --	N/A S/N	No	D: -- Hrs: -- C: --	/
77-12-06 R(2) 12/21/1977	[Recurring] Superseded by 2002-09-08	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	/
75-07-05 05/01/1977	Superseded by 77-12-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	/
74-15-02 01/01/1974	Superseded by 77-12-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	/
73-10-03 01/01/1973	Superseded by 77-12-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	/
70-02-01 01/01/1970	Superseded by 73-10-03	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	/
70-16-03 R 01/01/1970	Superseded by 77-12-06	-- Hrs: -- C: --	Superseded	No	D: -- Hrs: -- C: --	/


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Category: Appliance
 Manufacturer: Slick
 Model: 6361

Position: left
 P/N:
 S/N: 06100638

ATP Revision: 11/27/2019

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		Hrs: -- C: --			D: -- Hrs: -- C: --	

Category: Appliance

Position: right

ATP Revision: 11/27/2019

Manufacturer: Slick

P/N:

Model: 6361

S/N: 06031623

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		Hrs: -- C: --			D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Precision Airmotive

P/N: 2576554-2

Model: RSA-10ED1

S/N: 70931009

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2009-02-03 02/09/2009	[Recurring] To prevent a lean running engine, which could result in a substantial loss of engine power and subsequent,contd.	01/15/2019 Hrs: -- C: --	C/W at component O/H. NFA required	Yes	D: -- Hrs: -- C: --	
2008-08-14 04/29/2008	[Recurring] Superseded by 2009-02-03	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	
2008-06-51 E 03/12/2008	[Recurring] Superseded by 2008-08-14	-- Hrs: -- C: --	Superseded	Yes	D: -- Hrs: -- C: --	

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TABLE SIGNED*

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
--		Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Champion Aerospace

P/N: CH48110-1

Model: Oil Filters

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
--		Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Donaldson

P/N:

Model: P151936

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
--		Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Kelly Aerospace

P/N:

Model: 46C19836J

S/N: 11L00017

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Lear-Romec

P/N:

Model: RG 9480

S/N: E380A

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Champion Aerospace

P/N:

Model: REM32E

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	

Category: Appliance

Position:

ATP Revision: 11/27/2019

Manufacturer: Tempest

P/N:

Model: REM32E

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
--		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	